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APPLICATION NUMBER: 60/562,883

FILING DATE: *April 16, 2004*

RELATED PCT APPLICATION NUMBER: PCT/US04/33089

Certified by



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PROVISIONAL APPLICATION FOR PATENT COVER SHEET

This is a request for filing a PROVISIONAL APPLICATION FOR PATENT under 37 CFR 1.53(c).

Express Mail Label No. ET730095385US

INVENTOR(S)

Given Name (first and middle (if any))	Family Name or Surname	Residence (City and either State or Foreign Country)
Michael E. Gordon P.	Nanna Bierwagen	Moorhead, Minnesota Fargo, North Dakota

☐ Additional inventors are being named on the _____ separately numbered sheets attached hereto
TITLE OF THE INVENTION (500 characters max)**MAGNESIUM RICH COATINGS**Direct all correspondence to: **CORRESPONDENCE ADDRESS**
☐ Customer Number

OR

☒ Firm or Individual Name **Peter Rogalskyj, Esq.**
Address **Rogalskyj & Weyand, LLP**Address **P.O. Box 44**City **Livonia** State **New York** ZIP **14487-0044**Country **United States** Telephone **585-346-1004** Fax **585-346-1001****ENCLOSED APPLICATION PARTS (check all that apply)**

<input checked="" type="checkbox"/> Specification	Number of Pages 47	<input type="checkbox"/> CD(s), Number <input type="text"/>
<input type="checkbox"/> Drawing(s)	Number of Sheets 4	<input type="checkbox"/> Other (specify) <input type="text"/>
<input type="checkbox"/> Application Data Sheet. See 37 CFR 1.76		

METHOD OF PAYMENT OF FILING FEES FOR THIS PROVISIONAL APPLICATION FOR PATENT

<input checked="" type="checkbox"/> Applicant claims small entity status. See 37 CFR 1.27.	FILING FEE AMOUNT (\$) \$80.00
<input type="checkbox"/> A check or money order is enclosed to cover the filing fees	
<input checked="" type="checkbox"/> The Director is hereby authorized to charge filing fees or credit any overpayment to Deposit Account Number 50-0772	
<input type="checkbox"/> Payment by credit card. Form PTO-2038 is attached.	

The invention was made by an agency of the United States Government or under a contract with an agency of the United States Government.

☐ No.
☒ Yes, the name of the U.S. Government agency and the Government contract number are: **U.S. Air Force Office of Scientific Research Grant No. F49620-99-1-0283**

Respectfully submitted,

SIGNATURE

Date

04/16/2004

TYPED or PRINTED NAME

Peter Rogalskyj

REGISTRATION NO.

38,601

(if appropriate)

Docket Number:

047.00061 (RFT-118)

TELEPHONE

585-346-1004**USE ONLY FOR FILING A PROVISIONAL APPLICATION FOR PATENT**

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Applicant(s): Nanna et al.

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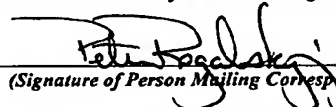
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Invention: **MAGNESIUM RICH COATINGS**

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CERTIFICATE OF MAILING BY "EXPRESS MAIL" (37 CFR 1.10)Applicant(s): **Nanna et al.**

Docket No.

047.00061 (RFT-118)

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Invention: **MAGNESIUM RICH COATINGS**

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TITLE: MAGNESIUM RICH COATINGS

**INVENTORS: Michael E. Nanna
 Gordon P. Bierwagen**

DOCKET NO.: 047.00061 (RFT-118)

MAGNESIUM RICH COATINGS

5

The present application supplements the disclosure set forth in U.S. Provisional Patent Application Serial No. 60/519,681, filed November 13, 10 2003, which provisional patent application, in its entirety, is hereby incorporated by reference.

The present invention was made with the support of the U.S. Air Force Office of Scientific Research Grant No. F49620-99-1-0283. The Federal Government may have 15 certain rights in this invention.

FIELD OF THE INVENTION

The present invention is directed, generally, 20 to a composition and method for its use in preventing or inhibiting corrosion of metals, particularly aluminum and aluminum alloys.

BACKGROUND OF THE INVENTION

25

Many metals are susceptible to corrosion. In this regard, atmospheric corrosion is of particular concern. Such corrosion may affect the performance and/or appearance of the metals affected, and the 30 products produced therefrom. In addition, when polymer coatings such as paints, adhesives or sealants are applied to the metal, corrosion of the underlying metal may cause a loss of adhesion between the polymer coating

and the base metal. A loss of adhesion between the polymer coating and the base metal may similarly lead to corrosion of the metal. Aluminum and aluminum alloys frequently require corrosion protection and improvements
5 in adhesion between the base aluminum (or aluminum alloys) and subsequent polymer coatings. Aluminum alloys, in particular, can be susceptible to corrosion since the alloying elements used to improve the metal's mechanical properties may decrease corrosion resistance.

10 Traditionally, precipitation grade hardened high strength Al alloys, containing heterogeneous microstructures formed from intermetallic compounds of Cu, Mg, Fe and Mn, are used with protective coating systems containing CrIV, chromates, and dichromates that
15 are especially effective at inhibiting Al alloy corrosion. These corrosion resistant aircraft coating systems for aluminum substrates typically consist of a conversion coating layer, a primer layer, and a topcoat. In practice, chromate conversion coatings ("CCC") can
20 either be deposited on the Al surface anodically by an applied anodic current, or developed chemically by the reaction of a tri- and hexa-valent chromium salt solution (i.e., Alodine™) with the Al metal. In spite of the widespread success and use of CCC systems in protecting
25 Al aircraft structures, as well as Cr-pigmentation, the use of chromates is being curtailed as they have been found to be carcinogenic, to be expensive to handle, and to the source of one of the highest airplane maintenance costs.

30 Generally, corrosion processes describe the oxidation of a metal at its surface which acts to weaken and/or disfigure it. Most metals are active enough to be converted to their oxides, and it is generally accepted

that corrosion occurs by an electrochemical action involving the creation of small galvanic cells on the surface of the metal. It has been observed that the bulk of structural corrosion damage to aging aircraft emanates from components involved in the joining process to the airframe itself, such as rivets, fasteners, lap splices, joints, and spot welding. All of these joining methods are associated with metallurgical and environmentally induced factors that affect the alloying elements in the metal, and, once changed, the exterior and interior surfaces of the aircraft become more susceptible to corrosion. For example, in the one fleet of aircraft, crevice corrosion occurring in the spot welded lap joint/doubler and environmentally induced corrosion around steel fasteners on the upper wing skins have been observed and addressed as major corrosion issues.

Categorically, there are three broad factors associated with corrosion processes in aluminum alloys: 1) metallurgical; 2) mechanical, and 3) environmental. Metallurgically induced factors include heat treatment, chemical composition of the alloying elements, material discontinuities, for example the presence of voids, precipitates, grain boundaries/orientation, and/or copper concentration in second-phase (S-phase). Mechanical factors include cycle-dependent fatigue and fatigue crack initiation. Further, environmental factors contributing to corrosion include temperature, moisture content, pH, electrolyte, type of salt present, and frequency and duration of exposure.

The most widely accepted factors contributing to corrosion processes in military aluminum aircraft structures, are direct chemical attack (e.g., aggressive phosphate ester hydraulic fluid leaks), galvanic

corrosion (e.g., when metals of different electrochemical potential are in contact in a corrosive medium), crevice corrosion (e.g., when a corrosive liquid gains access to crevices in or between components), pitting corrosion
5 (e.g., a localized attack that leads to the formation of deep and narrow cavities), and stress corrosion (e.g., when tensile stress or critical environment conditions cause dealloying to occur at grain boundaries which results in the formation of anodic precipitate areas).
10 Overall, among all of these corrosion types, material thinning by pitting at particle sites is the most basic corrosion mechanism affecting Al 2024 T-3 fuselage skin material.

Generally, Al 2024 T-3 is used for the exterior
15 fuselage, wing skins, and flight control surfaces, where pits are observed to form in exposed grain structure when subjected to environmental conditions that favor pitting. Pitting corrosion in Al 2024 T-3 occurs when cathodic particles (Al, Cu, Fe and Mn) dissolve in the alloy
20 matrix while anodic particles (Al and Mg) also dissolve, leading to intergranular corrosion. It has been approximated that there are roughly three times more anodic particles than cathodic particles in Al 2024 T-3, and, therefore, it is prone to intergranular corrosion
25 induced by pitting. The failure to address airframe corrosion damage due to shallow pitting, or damage related to fatigue and cracking can lead to catastrophic consequences, such as, incidents in the mid 1950s when two Comet airplanes failed in high altitude flight and
30 the Aloha Airlines incident of 1988.

As mentioned above, prior art techniques for improving corrosion resistance of metals widely employ the use of chromate conversion coatings to passivate the

surface. Such chromate treatments are undesirable, however, because the chromium used is highly toxic, carcinogenic, and environmentally undesirable. Phosphate conversion coatings are also used, but generally provide
5 substantially less corrosion protection unless used in conjunction with a chromate.

Recently, various techniques for eliminating the use of chromates in corrosion inhibition and adhesion promotion treatments have been proposed. However, many
10 of these proposed techniques have been proven to be ineffective, or to require time consuming, energy inefficient, multi-step processes. Thus, there remains a need for a simple, low cost, effective technique for inhibiting corrosion of metals, particularly for aluminum
15 and aluminum alloys. The present invention, at least in part, is directed to meeting this need.

SUMMARY OF THE INVENTION

20 The present invention relates to a method of treating a metal to improve the metal's corrosion resistance. The method includes applying, to the surface of the metal, a coating which comprises magnesium powder and which is substantially free of chromium.

25 The present invention also relates to a coating composition that includes magnesium powder and a silane modified epoxy isocyanate hybrid polymer or prepolymer.

BRIEF DESCRIPTION OF THE DRAWINGS

30 Figures 1A-1C are schematic diagrams showing the development of a multi-layer covalent structure used in accordance with certain embodiments of the present

invention. Figure 1A shows a surface silation/amination layer. Figure 1B shows sequential reaction of this layer with HMDI or MDI. Figure 1C shows a further polymerization that is surmised to take place from the surface layer and extending into the bulk. Figure 1D shows a reaction of an isocyanate monomer and a crosslinker, 7-phenyl-1-[4-(trimethylsilyl)-butyl]-1,2,3,4-tetrahydroquinoxalin-6-ol depicting bulk epoxy/isocyanate interpenetrating network growth throughout the bulk phase.

Figure 2A is a graph of an open circuit potential SCE vs PVC for Mg-rich epoxy/polyamide primers at pH=6.2 in 3% NaCl. Figure 2B is a graph showing $|Z|$ modulus at 0.01 Hz vs PVC for Mg-rich epoxy/polyamide primers at pH=6.2 in 3% NaCl.

Figure 3A is a graph showing particle size distributions for Mg powders, Eckagranules™ PK31 and PK51. Figure 3B is a ternary mixture diagram for PK31, PK51, and Aerosil™ R202 showing regions corresponding to theoretically calculated CPVC's.

Figures 4A-4D are images showing results from a modified UL-94 flammability test for 50% PVC hybrid epoxy/urea/urethane N3300 (Figure 4A); 50% PVC hybrid E23A (Figure 4B); 50% PVC MC-PUR (Figure 4C); and 50% PVC epoxy-polyamide (Figure 4D).

Figures 5A-5D are images showing results from a Prohesion™ test conducted in accordance with ASTM D5894-96) for Al 2024 T-3 panels coated with the formulation set forth in Table 2, entry C (Mg-rich primer hybrid N3300 at 50% PVC with ELT™ topcoat at 0 hours (Figure 5A); at 1200 hours (Figure 5B); at 3000 hours (Figure 5C); and at 4800 hours (Figure 5D). Figure 5E is an image showing the result from a Prohesion™ test conducted

in accordance with ASTM D5894-96) for a Al 2024 T-3 panel coated with an epoxy Epon™828/Ancamide™ 2353 polyamide (a non-pigmented primer) with ELT™ topcoat at 1800 hours.

Figures 6A-6C are images of scribed Al 2024 T-3 samples after 11 days immersion and EIS testing at pH=12. The samples were coated with E23A (Figure 6A), MC-PUR (Figure 6B), or epoxy-Mannich base (Figure 6C).

Figures 7A-7C are images of scribed Al 2024 T-3 samples after 11 days immersion and EIS testing at pH=2.8. The samples were coated with E23A (Figure 7A), MC-PUR (Figure 7B), or epoxy-Mannich base (Figure 7C).

DETAILED DESCRIPTION OF THE INVENTION

The present invention relates to a method of treating a metal to improve the metal's corrosion resistance. The method includes applying, to the surface of the metal, a coating which comprises magnesium powder and which is substantially free of chromium.

A variety of metals, such as aluminum and aluminum alloys, can be protected using the methods of the present invention. Illustratively, the methods of the present invention can be used to treat aluminum alloys which contain copper (which is meant to include heterogeneous microstructures formed from intermetallic compounds containing copper and one or more other metals, such as Mg, Fe, and Mn. For example, the methods of the present invention can be used to treat copper-containing aluminum alloys which are commonly used in airplanes and other aircraft, such as Al 2024 T-3 and Al 7075 T-6.

As used herein, the phrase "improve the metal's corrosion resistance" is meant to be broadly construed

and can be ascertained by any suitable qualitative or quantitative method known to those skilled in the art. Illustratively, a metal's corrosion resistance can be determined by PROHESION™ exposure, for example, in accordance with ASTM D5894-96, which is hereby incorporated by reference. Any increase in the metal's corrosion resistance is to be deemed to "improve" its corrosion resistance. Increases in corrosion resistance can be determined, for example, visibly by comparing test samples coated in accordance with the method of the present invention to uncoated test samples or to test samples coated only with topcoat. As indicated above, the level of corrosion resistance can be ascertained qualitatively, as by the visual observation of blistering, peeling, curling, bubbling, or other indicia of coating failure or delamination or by the visual observation of pitting and other indicia of corrosion of the metal. Such observations can be made at a single point in time (e.g., after PROHESION™ exposure in accordance with ASTM D5894-96 for about 1800 hours, about 2000 hours, about 2500 hours, about 3000 hours, about 3500 hours, about 4000 hours, about 4500 hours, about 5000 hours, etc.), or they can be made over a period of time.

As discussed above, the method of the present invention is carried out by applying, to the surface of the metal, a coating which comprises magnesium powder and which is substantially free of chromium.

"Magnesium powder", as used herein is meant to refer to a collection of micron-sized particles (e.g., particles having a diameter of about 1-1000 microns, such as of about 10-100 microns, etc.) which contain magnesium metal and/or the oxides thereof. For example, particles which make up the magnesium powder can include a

magnesium metal core and coating of magnesium oxide on the surface thereof. It should be noted, in this regard, that reference here to "diameter" is not to imply that the particles which make up the magnesium powder are
5 necessarily spherical; the particles can be spherical, ellipsoidal, cubical, rod-shaped, disk-shaped, prism-shaped, etc., and combinations thereof; and in the case where a particle is other than spherical, the "diameter" referred to above is meant to refer to the diameter of a
10 hypothetical sphere having a volume equal to that of the particle. The particles contained in the magnesium powder can be of substantially uniform particle size or not. For example, the magnesium powder can include a mixture of two or more magnesium particle powders, each
15 having different mean particle size distributions, such as in the case where the magnesium powder includes a first magnesium particle powder and a second magnesium particle powder, where the first magnesium particle powder and a second magnesium particle powder have
20 substantially different mean particle size distributions. As used in this context, two mean particle size distributions, X and Y, are to be deemed to be "substantially different" when either the ratio of X:Y or the ratio Y:X is greater than about 1.5, such as greater
25 than about 1.6, greater than about 1.7, greater than about 1.6, greater than about 1.8, greater than about 1.9, greater than about 2, from about 1.1 to about 4, from about 1.5 to about 3, from about 2 to about 2.5, from about 2.1 to about 2.5, and/or from about 2.2 to
30 about 2.4. Additionally or alternatively, the first magnesium particle powder and the second magnesium

particle powder can be selected such that the mixture's bulk density is greater than the first magnesium particle powder's bulk density and such that the mixture's bulk density is greater than the second magnesium particle powder's bulk density, for example, as in the case where the mixture's bulk density is at least about 2% greater (e.g., at least about 5% greater, at least about 8% greater, etc.) than the first magnesium particle powder's bulk density and such that the mixture's bulk density is at least about 2% greater (e.g., at least about 5% greater, at least about 8% greater, etc.) than the second magnesium particle powder's bulk density.

As further illustration, the magnesium powder used in the practice of the present invention can include a mixture of a first magnesium particle powder having a mean particle size distribution of from about $25\mu\text{m}$ to about $35\mu\text{m}$ (such as in the case where first magnesium particle powder has a mean particle size distribution of from about $27\mu\text{m}$ to about $33\mu\text{m}$ and/or as in the case where the first magnesium particle powder has a mean particle size distribution of about $30\mu\text{m}$) and a second magnesium particle powder having a mean particle size distribution of from about $65\mu\text{m}$ to about $75\mu\text{m}$ (such as in the case where second magnesium particle powder has a mean particle size distribution of from about $67\mu\text{m}$ to about $73\mu\text{m}$ and/or as in the case where the second magnesium particle powder has a mean particle size distribution of about $70\mu\text{m}$).

As yet further illustration, the magnesium powder used in the practice of the present invention can include a mixture of a first magnesium particle powder having a mean particle size distribution of from about $25\mu\text{m}$ to about $35\mu\text{m}$ and a second magnesium particle powder

having a mean particle size distribution of from about 65 μ m to about 75 μ m, where the volume ratio of first magnesium particle powder to second magnesium particle powder is from about 40:60 to about 60:40, for example, 5 as in the case where the volume ratio of first magnesium particle powder to second magnesium particle powder is from about 45:55 to about 55:45; as in the case where the volume ratio of first magnesium particle powder to second magnesium particle powder is from about 50:50 to about 10 55:45; and/or as in the case where the volume ratio of first magnesium particle powder to second magnesium particle powder is about 58:42.

As yet further illustration, the magnesium powder used in the practice of the present invention can 15 include a mixture of a first magnesium particle powder having a mean particle size distribution of about 30 μ m and a second magnesium particle powder having a mean particle size distribution of about 70 μ m, where the volume ratio of first magnesium particle powder to second 20 magnesium particle powder is from about 40:60 to about 60:40, for example, as in the case where the volume ratio of first magnesium particle powder to second magnesium particle powder is from about 45:55 to about 55:45; as in the case where the volume ratio of first magnesium 25 particle powder to second magnesium particle powder is from about 50:50 to about 55:45; and/or as in the case where the volume ratio of first magnesium particle powder to second magnesium particle powder is about 58:42.

As discussed above, the method of the present 30 invention is carried out by using a coating which (i) includes the aforementioned magnesium powder and (ii) is substantially free of chromium. As used herein, the coating is to be deemed to be substantially free of

chromium if the ratio of the weight of chromium metal or ion in the coating to the weight of magnesium metal or ion in the coating is less than 20%, such as less than about 18%, less than about 15%, less than about 12%, less than about 10%, less than about 5%, less than about 2%, less than about 1%, less than about 0.5%, and/or about zero.

Aside from the coating containing magnesium powder and being substantially free of chromium, coating can contain other substances, such as binders, solvents, and the like.

Illustratively, the coating can further (i.e., in addition to the magnesium powder) include a binder. The binder can be, for example, any suitable polymeric material (e.g., a polymer or copolymer) or a prepolymer (e.g., a monomer or oligomer) or combination of prepolymers which, upon polymerization or copolymerization, forms a polymer or copolymer. Illustratively, the binder can include a hybrid polymeric matrix or a plurality of hybrid polymeric matrices or other polymer composites or alloys that contain a polymer backbone with at least two types of reactive groups that can take part in crosslinking and network formation under at least two different mechanisms; and/or the binder can contain a prepolymer or combination of prepolymers which, upon polymerization or copolymerization, forms the aforementioned hybrid polymeric matrix, hybrid polymeric matrices, or other polymer composites or alloys.

For example, in one embodiment of the method of the present invention, the binder includes a polyisocyanate prepolymer and an epoxy prepolymer.

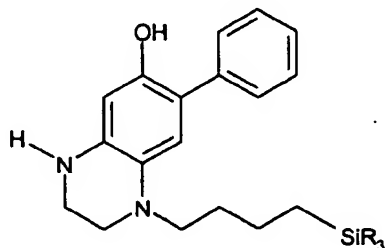
Suitable polyisocyanate prepolymers include, for example, aliphatic polyisocyanate prepolymers, such

as 1,6-hexamethylene diisocyanate homopolymer ("HMDI") trimer, and aromatic polyisocyanate prepolymers, such as 4,4'-methylenediphenylisocyanate ("MDI") prepolymer. Combinations of two or more aliphatic polyisocyanate
5 prepolymers, combinations of two or more aromatic polyisocyanate prepolymers, and/or combinations of one or more aliphatic polyisocyanate prepolymers and one or more aromatic polyisocyanate prepolymers can also be employed.

Suitable epoxy prepolymers include, for
10 example, any common epoxy resin, such as those which include at least one multifunctional epoxy resin (i.e., an epoxy resin having two or more epoxy groups per molecule). Examples of such epoxy resins include polyglycidyl ethers of pyrocatechol, resorcinol,
15 hydroquinone, 4,4'-dihydroxydiphenyl methane (or bisphenol F, such as RE-404-S or RE-410-S available commercially from Nippon Kayaku, Japan), 4,4'-dihydroxy-3,3'-dimethyldiphenyl methane, 4,4'-dihydroxydiphenyl dimethyl methane (or bisphenol A), 4,4'-dihydroxydiphenyl
20 methyl methane, 4,4'-dihydroxydiphenyl cyclohexane, 4,4'-dihydroxy-3,3'-dimethyldiphenyl propane, 4,4'-dihydroxydiphenyl sulfone, and tris(4-hydroxyphenyl)methane; polyglycidyl ethers of transition metal complex chlorination and bromination
25 products of the above-mentioned diphenols; polyglycidyl ethers of novolacs; polyglycidyl ethers of diphenols obtained by esterifying ethers of diphenols obtained by esterifying salts of an aromatic hydrocarboxylic acid with a dihaloalkane or dihalogen dialkyl ether;
30 polyglycidyl ethers of polyphenols obtained by condensing phenols and long-chain halogen paraffins containing at least two halogen atoms; N,N'-diglycidyl-aniline; N,N'-dimethyl-N,N'-diglycidyl-4,4'-diaminodiphenyl

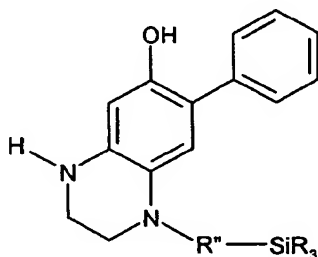
methane; N,N,N',N'-tetraglycidyl-4,4'-diaminodiphenyl
methane; N,N'-diglycidyl-4-aminophenyl glycidyl ether;
N,N,N',N'-tetraglycidyl-1,3-propylene
bis-4-aminobenzoate; phenol novolac epoxy resin; cresol
5 novolac epoxy resin; and combinations thereof. Among the
commercially available epoxy resins suitable for use
herein are polyglycidyl derivatives of phenolic
compounds, such as those available under the tradenames
EPON 828, EPON 1001, EPON 1009, and EPON 1031, from Shell
10 Chemical Co.; DER 331, DER 332, DER 334, and DER 542 from
Dow Chemical Co.; GY285 from Ciba Specialty Chemicals,
Tarrytown, N.Y.; and BREN-S from Nippon Kayaku, Japan.
Combinations of the aforementioned epoxy prepolymers and
other epoxy prepolymers can also be employed.
15 Monofunctional epoxy resins, can also be used, for
example, as a reactive diluent or crosslink density
modifier.

The method of the present invention can also
include contacting the binder with a crosslinker.
20 Suitable crosslinkers include, for example, silanated
tetrahydroquinoxalins, such as 7-phenyl-1-[4-
(trialkylsilyl)-butyl]-1,2,3,4-tetrahydroquinoxalin-6-ol
and other 7-phenyl-1-[4-(trialkylsilyl)-alkyl]-1,2,3,4-
tetrahydroquinoxalin-6-ols. As used herein, 7-phenyl-1-
25 [4-(trialkylsilyl)-butyl]-1,2,3,4-tetrahydroquinoxalin-6-
ol is meant to include compounds having the following
Formula I:



wherein each R is the same or different and represents an alkoxy group. "Alkoxy", as used in this context, is meant to refer to a group having the formula -OR' where R' is a substituted or unsubstituted alkyl (e.g., methyl, ethyl, n-propyl, isopropyl, n-butyl, isobutyl, t-butyl or other C1-C6 alkyl or other lower alkyl) or a substituted or unsubstituted aryl group. As used herein, 7-phenyl-1-[4-(trialkylsilyl)-alkyl]-1,2,3,4-tetrahydroquinoxalin-6-ol is meant to include compounds having the following

10 Formula II:



wherein each R is as defined above with regard to Formula I and wherein R" is a propylene moiety (e.g., -CH₂CH₂CH₂-), a butylene moiety (e.g., -CH₂CH₂CH₂CH₂-), a pentylene moiety (e.g., -CH₂CH₂CH₂CH₂CH₂-), or other C1-C6 alkylene or other lower alkylene-containing bridging moieties.

15

The step of contacting the binder and crosslinker can be carried out, for example, prior to, or simultaneously with the step of applying the coating to the surface of the metal. Illustratively, the crosslinker can be combined with the binder in the coating formulation, and the coating formulation (containing crosslinker, magnesium powder, binder, etc.) can be applied in a single step. Alternatively, the crosslinker can be applied to the metal surface prior to applying the coating formulation (containing magnesium powder, binder, etc.). Still alternatively, the

20

25

crosslinker can be applied to the metal surface prior to applying the coating formulation, and the coating formulation can contain additional crosslinker (in addition to containing magnesium powder, binder, etc.).

5 Whether or not a silanated tetrahydroquinoxal-
inol crosslinker or other crosslinker is employed, the
method of the present invention can further include the
step of contacting the metal surface with an amine-
containing organo-silane prior to the aforementioned step
10 of applying a coating to the surface of the metal. A
variety of amine-containing organo-silanes can be
employed in this regard, such as amine-containing organo-
silanes having the formula $H_2N-R^a-NH-R^b-S(OR^c)_3$, where each
of R^a and R^b independently represents a methylene moiety
15 (e.g., $-CH_2-$), an ethylene moiety (e.g., $-CH_2CH_2-$), a
propylene moiety (e.g., $-CH_2CH_2CH_2-$), a butylene moiety
(e.g., $-CH_2CH_2CH_2CH_2-$), a pentylene moiety (e.g.,
 $-CH_2CH_2CH_2CH_2CH_2-$), or other C1-C6 alkylene or other lower
alkylene-containing bridging moieties; and where R^c is the
20 same or different and represents an alkoxy group.
Illustratively, suitable amine-containing organo-silanes
include (N- β -(aminoethyl)- γ -aminopropyltrimethoxysilane,
(N- β -(aminoethyl)- γ -aminopropyltriethoxysilane, (N- β -
(aminoethyl)- γ -aminobutyltrimethoxysilane, and (N- β -
25 (aminoethyl)- γ -aminobutyltriethoxysilane.

When used together in the method of the present
invention, it is believed that the amine-containing
organo-silane, polyisocyanate prepolymers, epoxy
prepolymers, and silanated tetrahydroquinoxalinol
30 crosslinkers form a silane modified epoxy isocyanate
hybrid binder which is bonded to the metal surface. For
example, initial application of an amine-containing
organo-silane can result in a surface silation amination

layer, as shown in Figure 1A. This layer is then reacted with the polyisocyanate prepolymers, such as HMDI and MDI, as shown in Figure 1B; and, as shown in Figure 1C, further polymerization is surmised to take place from the surface layer and extend into the bulk. More particularly, hydrolysis (e.g., with atmospheric or other ambient moisture) of the isocyanate can form carbamic acid (forming both amines and carbon dioxide that are intermediates in a Hoffmann rearrangement of an isocyanate with water) to yield a reactive/reaminated layer and a covalent structure with hydrogen bonded parallel multi layers. The bulk reaction, extending from the surface occurs between (i) the polyisocyanate in the binder formulation and the aminated surface (e.g., as discussed above and shown in Figure 1C) and (ii) the polyisocyanate and the silanated tetrahydroquinoxalinol crosslinker (for example, as shown in Figure 1D) which is surmised upon further hydrolysis to form both polyurea and polysiloxane interpenetrating network ("IPN") structure. In this manner, for example, a silane modified epoxy isocyanate hybrid binder material can be prepared from a polymeric material consisting of polyurea, polyurethane (from polyisocyanate prepolymer), epoxy-amine, and organo-silane linkages.

As discussed above, the method of the present invention is carried out by applying the coating discussed above to the surface of the metal whose corrosion resistance is to be improved.

The coating can be applied in the form of a suspension, dispersion, or solution in a suitable solvent or combination of solvents, examples of which include polypropylene carbonate, ethyl-3-ethoxypropionate ("EEP"), and combinations thereof. Application can be

carried, for example, by any suitable technique, such as spraying (e.g., airless spraying or spraying with the use of air), brushing, rolling, flooding, immersion, etc., to achieve a suitable coating thickness, such as from about
5 10 to about 100 microns, from about 30 to about 80 microns, from about 40 to about 60 microns, and/or about 50 microns. As discussed above, the coating can be applied directly to the metal's surface, or it can be applied indirectly to the metal's surface, such as in the
10 case where the metal's surface is first contacted with an amine-containing organo-silane prior to the application of the coating. Where an amine-containing organo-silane is employed, contact between the amine-containing organo-silane and the metal surface can be carried out, for
15 example, using any suitable technique, such as spraying (e.g. airless or otherwise), brushing, rolling, flooding, immersion, and the like, for example, as described in Walker, "Organo Silanes as Adhesion Promoters for Organic Coatings," Journal of Coatings Technology, 52(670): 49-61
20 (1980), which is hereby incorporated by reference.

Once applied, the coating can be cured, for example, for from about 1 week to about 1 month, such as for about two weeks, at a temperature of from about room temperature to about 50°C, such as at from about 30°C to
25 about 40°C and/or at about 35°C.

The coating can be top coated using any compatible topcoat formulation, such as Extended Lifetime™ Topcoat, for example by spraying or brushing to achieve a topcoat thickness of from about 20 to about 200
30 microns, such as from about 50 to about 150 microns, from about 80 to about 120 microns, and/or about 100 microns.

The inventors have found that corrosion resistance (as determined by PROHESION™ exposure in

accordance with ASTM D5894-96, which is hereby incorporated by reference) in excess of 3,000 hours on 2024 T-3 aluminum alloy can be achieved with the methods of the present invention.

5 The coating can include, in addition to magnesium powder, other materials. In one embodiment, the coating further includes a rare earth metal, such as cerium. The cerium can be present in the form of cerium metal, cerium oxides, cerium salts, or combinations
10 thereof. The cerium can be applied to the magnesium powder or a portion thereof, for example in the form cerium nitrate or other cerium salt. For example, in cases where the magnesium powder includes a mixture of two or more magnesium particle powders, each having
15 different mean particle size distributions, such as in the case where the magnesium powder includes a first magnesium particle powder and a second magnesium particle powder, where the first magnesium particle powder's mean particle size distributions is less than $20\mu\text{m}$ and where
20 the second magnesium particle powder's mean particle size distributions is greater than $20\mu\text{m}$, the cerium can be applied to the surface of the first magnesium particle powder but not to the second magnesium particle powder. Additionally, or alternatively, the cerium metal, oxide,
25 or salt can be dispersed in a binder used in the coating. Still additionally or alternatively, the cerium metal, oxide, or salt can be applied to the metal surface, e.g., in the form of cerium nanoparticles, prior to applying the coating thereto.

30 Certain aspects of the present invention are further illustrated with the following examples.

EXAMPLES

Example 1 -- Initial Electrochemical and Exposure Studies

5 A. Open Circuit Potential and Electrochemical Impedance Spectroscopy EIS Studies

The electrochemical studies of Mg-rich primers (without topcoat) formulated in our laboratory were first carried out on the surface of the primed Al 2024 T-3 alloy immersed in 3% NaCl solution. The corrosion potential, E_{corr} , or open circuit potential (OCP), for the coatings in contact with the alloy was monitored and the (EIS) spectra of three primer sets, as formulated in an epoxy-polyamide polymer matrix, were recorded as a function of time. The OCP is the mixed potential achieved when a corrosion reaction is occurring between the anode and cathode of the reaction system. The data presented in Figures 2A and 2B are for three of these primers based on Eckart (Eckagranules™) ~ 50 micron average particle size distribution (PSD) Mg powder at 43, 46 and 50% PVC. These data indicate that the most effective protection from just the primer is about 46% PVC, which was the estimated CPVC for this system.

Figure 2A gives (OCP) vs. exposure time for Mg-rich primers, formulated at 43, 46, and 50% PVC in a polyamide/epoxy coating polymer exposed to 3% NaCl solution at pH ~ 6.2. Interpretation of the events is as follows. Initial OCP values for the three sets correspond to a single electron transfer potential for Mg metal, $E_{\text{Mg}} = -1.50\text{V}$ to $-1.60\text{ V}_{\text{SCE}}$, and the primers appear to be acting like bare Mg. Subsequently, over a 24 hour period, Mg and the Al alloy polarize to a mixed potential corresponding to the corrosion potential, E_{corr} , at which

the Mg is still sacrificially protecting the Al 2024 whose E_{2024} is -0.68 V vs. saturated calomel electrode (SCE). The observed mixed potential for Mg and Al alloy in 3% NaCl was found to be about $E_{\text{corr}} = -0.90$ V to -1.00 V_{SCE}. OCP values extending beyond the initial 24-hour period varied according to primer PVC. The initial lower mixed potential value, E_{corr} (Figure 2A), for the 43% PVC sample is thought to be due to the lower effective active metal area as a result of higher polymer coverage at the Mg/Al alloy interface. Initially, the Mg-anode dominates the OCP. The gradual rise in OCP for the 43% PVC sample toward $E_{2024} = -0.68$ mV_{SCE} is assumed to be due to reactive consumption of the exposed Mg in this system and the disbonding of epoxy coating polymer from the cathode surface. The gradual decrease in OCP of the 43% PVC sample toward $E_{\text{mix}} = -0.90$ V to -1.00 V_{SCE} may be due to resistance polarization by the formation and packing of Mg oxides in the coating. The initial and continuous decrease in OCP of the 50 PVC sample is concluded to be due to a higher void volume in the primer as well as a higher cathode area at the primer alloy interface. The OCP of the 46% PVC sample quickly arrives at the $E_{\text{mix}} = -0.90$ V to -1.00 V_{SCE} value and remains constant for the duration of the test time period. Thus, it is surmised that the 46% PVC primer corresponds to the critical pigment volume concentration (CPVC) for the primer, suggesting that cathodic protection of the Al alloy due to Mg metal occurs most effectively at or near CPVC. Figure 2B shows the impedance modulus $|Z|$ versus exposure time measured in 3% NaCl solution at pH = 6.2 on 43%, 46%, and 50% PVC Mg-rich primers. This figure demonstrates the effect of PVC at CPVC for Mg-rich primers. The Z modulus values for the 46% PVC samples

yielded a higher impedance modulus over the 28-day period suggesting proper formulation at or near the critical pigment volume concentration, which is required to ensure close packing of Mg pigment with minimum resistance from the polymer matrix of the system, but with polymer matrix content sufficient enough to ensure good substrate wetting and reasonable physical properties from the primer.

Trends in OCP data suggest three distinct periods that distinguish the evolution and effectiveness of cathodic protection in the Mg-rich primers as a function of exposure time. These are as follows.

Period I. Initial immersion day one, the "activation" period when the value of the corrosion potential shifts to a cathodic value $-1.1V$ vs SCE, corresponding to the Mg metal/Al 2024 T-3 mixed potential in the electrolyte. Magnesium immediately begins to react with the sodium-chloride solution; it is "activated" leading to better metal-to-metal electrical contact being established between magnesium particles and Al surface.

Period II. Once initially past the "activation" period, the cathodic protection mechanism reaches its peak due to a maximum in the ratio of magnesium-to-aluminum area ratio. This occurs around day 5-7 when the corrosion potential shifts to a more anodic value of about $-0.9V$ vs SCE; it is where a relative stabilization called the "transition" period occurs.

Period III. After the transition period, and up to day 21, the corrosion potential shifts out of the cathodic protection domain, and the potential fluctuates as the film's solution chemistry begins to change. At this time, oxygen reduction begins to occur on the upper

part of the film causing a local increase in pH that changes the corrosion products from magnesium hydroxychlorides to magnesium hydroxides, the same as at the interface.

5

B. Initial Accelerated Test

In metal-rich coatings, the pigment volume concentration (PVC), is high and close to critical PVC, in the region at which paint properties such as water permeability and cohesive strength change dramatically. Therefore, Prohesion™ cyclic exposure in dilute Harrison's solution with no topcoat allows easy access of acidic electrolyte, atmospheric oxygen, CO₂ and water to the coating's Mg anode. Topcoating the Mg-rich primer insulates or screens it from the primary cyclic effects of Prohesion™ which in turn prevents observation of the processes that occur when the coating is scratched or chinked when the Al alloy is exposed to an acid rain environment. In order to better observe the occurrence of such processes, primed panels were directly exposed to dilute Harrison's solution, without topcoat, and monitored. The pH of dilute Harrison solution is about -4.5 which corresponds to the pH at which Mg metal readily forms salts with CO₂, SO₄²⁻ and OH⁻. The formation of these salts was observed to occur on the surface and at the interface of the primer over a given time interval that corresponds to three distinct events:

1. EDXA spectra have revealed formation of magnesium carbonate hydrates at the primer liquid/vapor interface, dypingite [Mg₅(CO₃)₄(OH)₂·8H₂O] and hydromagnesite [Mg₅(CO₃)₄(OH)₂·4H₂O]. These salts have been observed only to be present up to the first 500

hours of exposure for all non top-coated primed Mg-rich panels tested.

2. For exposure times beyond 500-hours, brucite $[Mg(OH)_2]$ domains begin to form and subsequently
5 extend throughout the bulk of the primer. During this time the aluminum alloy remains cathodically protected as scribed lines remain unblemished.

3. For exposure times greater than 1,300 hours, primer failure and film delamination correspond to
10 the accumulation of hexahydrate $[(MgSO_4) \cdot 6 H_2O]$ compound at the interface. Failure occurs when Mg-metal and brucite structure have been depleted from the coating polymer matrix and sufficient hexahydrate salts have accumulated at the alloy interface, at which time the
15 coating polymer ruptures and fragments from compressive forces exerted by hexahydrate structures.

The first 24 hours of exposure to salt fog solution with atmospheric CO_2 , magnesium forms magnesium carbonate compound, at the surface, $Mg_5(OH)_5 \cdot CO_3$, which is
20 replaced by a more densely packed magnesium hydroxide $Mg(OH)_2$ pseudo-hexagonal crystal structure. The rosette structure observed in the magnesium epoxy primer scanning electron microscopy (SEM) images is consistent with brucite magnesium hydroxide $Mg(OH)_2$ (acicular needle)
25 crystal formed in Prohesion™ exposure. Further observations were made on the Mg-rich primers exposed to Prohesion™ cyclic salt fog with dilute Harrison's solution. First, white oxide area magnesium hydroxide (brucite) formed over magnesium metal and energy
30 dispersed X-ray analysis (EDXA) measurements indicated the presence of magnesium, oxygen and aluminum with a minimum amount of carbon detected. Second, in the scribed area with no epoxy matrix or magnesium metal

originally present, EDXA spectra show carbon, oxygen, magnesium and aluminum with possible presence of dypingite (magnesium carbonate) structure over the exposed aluminum surface.

5 In summary, Table 1 gives the relative pH, the solubility product, and the water solubility for magnesium salts identified in EDXA spectra. It was observed that the salts generated during the first 1,000 hours of exposure increased in local pH according to a
10 stratification scheme from the coating/alloy surface interface toward the external surface of the coating.

TABLE 1
Magnesium Salts Solubility pH

15

Salt	(designation)	Ksp	H ₂ O g/100ml	pH
Mg(OH) ₂	Brucite	7.1×10^{-12}	7.8×10^{-4}	9.6 ~10.4
MgCO ₃ ·5 H ₂ O	Magnesite	3.8×10^{-6}	0.002	8.0 ~ 8.8
MgSO ₄ ·6 H ₂ O	Hexahydrate	(soluble)	95	6 ~ 9

20 The degradation process of Mg-rich coatings exposed to an acidic environment may be described as follows. The more acidic salt (i.e., hexahydrate) was identified at the alloy interface where local pH
25 conditions are lower due to anodic polarization conditions that drive cationic species, such as Al³⁺ from solution towards the Al interface. The carbonate salt was found to develop on the top of the coating along with Mg(OH)₂, and both are identified as species that form at
30 higher pH. In addition, the damaged/scribed areas did not degrade neither the coating polymer nor the alloy surface until after the depletion of Mg(OH)₂, and as the accumulation of hexahydrate salts occurred. It has been

reported that aqueous magnesium hydroxide acts a pH buffer that does not exceed a pH=10.5, even in the presence of excess $Mg(OH)_2$.

5 C. Conclusions of Preliminary Feasibility Studies

These very encouraging results were obtained from a simple Mg-rich coating based on an off-the-shelf (OTS) polymer system with no optimization efforts. They showed that the oxidation products of the Mg pigment in
10 an exposure environment, fairly typical of what an actual system might see in field exposure, did not cause basic corrosion of the Al 2024 T-3 alloy. Further, the Mg-rich system did provide cathodic protection to the Al 2024 T-3, giving the system significant corrosion protection
15 properties in a completely Cr-free system with no chromates in pretreatment nor chromate pigments in primer. Our studies proceeded to improve the coating polymer system and additional formulation studies. These studies are described in Example 2.

20

Example 2 -- Formulation Improvement by Coating Polymer Design and Preparation

A. Coating Polymer Selection

25 Traditionally, two-pack zinc epoxy/polyamide polymer materials have been used for cathodic protection of steel as they result in crosslinked matrices with a good adhesion and resistance against alkalis, so that any alkaline reaction involving zinc does not affect the
30 binder itself (Van Vliet, Prog. Organic Coatings, 34:220-226 (1998), which is hereby incorporated by reference). More recently, epoxy siloxane "hybrid" coatings have been reported (Keijman, High Solids Coatings: Experience in

Europe and USA - Paper 40, Proceedings PCE Conference,
The Hague, The Netherlands "Protecting industrial and
marine structures with coatings", March 1997, which is
hereby incorporated by reference) to represent a
5 significant advancement compared to epoxy, epoxy acrylic
and polyurethane coatings. Hybrid polymeric matrices,
for high performance primers, are designed as polymer
composites or alloys that contain a polymer backbone with
at least two types of reactive groups that can take part
10 in crosslinking and network formation under at least two
different mechanisms.

B. Silane Modified Multi-Layer/IPN Polymer Matrix

The design of an improved polymeric matrix for
15 Mg-rich coatings involves an easy to prepare multi-layer
scheme that requires minimum preparation of the Al alloy
surface and is derived from existing sol-gel technology.
The reaction scheme entails initial application of an
organo-silane (N- β -(aminoethyl)- γ -aminopropyltrimethoxy-
20 silane) with subsequent grafting of organic layers from
the surface into the bulk by utilizing a novel silane
modified crosslinker. The coating scheme is akin to the
"sol-gel" process, but involves a multilayer approach
that utilizes an organo-silane substrate treatment from
25 which a moisture-cure polyisocyanate is applied which has
is believed to involve an initial reaction with water to
form an unstable carbamic acid intermediate that
spontaneously decarboxylates into an amine and carbon
dioxide, as shown in Figures 1A-1C. To complete the
30 scheme, further bulk crosslinking reactions between
epoxy, silanol, and isocyanate are proposed to occur with
from an aminated surface into the bulk by employing a
bulk/surface crosslinker. In brief, the prepared

organo-silane modified surface was subsequently sprayed with a 20 percent solution of polyisocyanate in propylene carbonate, with one of two poly isocyanate prepolymers, (i) 1,6-hexamethylene diisocyanate homopolymer (HMDI) trimer (as depicted in Figures 1A-1C) and (ii) 4,4'-methylenediphenylisocyanate (MDI) prepolymer similar scheme. Uniform coverage of the wet surface was approximated at ~ 2 mils (50 microns) using a wet film thickness gauge.

10 The bulk reaction, extending from the surface, occurs between (i) the polyisocyanate, in the primer formulation and the aminated surface (Figure 1C) and (ii) the isocyanate and 7-phenyl-1-[4-(trimethylsilyl)-butyl]-1,2,3,4-tetra-hydro-quinoxalin-6-ol crosslinker (Figure 15 1D) which is surmised upon further hydrolysis to form both polyurea and polysiloxane IPN structure. The silane modified epoxy (HMDI or MDI) hybrids results from a polymeric material consisting of polyurea, polyurethane (from polyisocyanate prepolymer), epoxy-amine, and 20 organo-silane linkages.

C. Coating Formulations

The materials used in this study are summarized in Table 2.

TABLE 2
Magnesium Rich Primer Materials

Formulation	Materials
(A) Hyb-E23A	Desmodur™ E23-A, Aerosil R202, Eckagranules™ PK51/31, Epon™ 1001CX, Propylene Carbonate/EEP
(B) MC-PUR*	Desmodur™ E23-A, Bentone® 34, Eckagranules™ PK51/31, Anti Terra®U, Aromatic solvent Xylol
(C) Hyb-N3300	Desmodur™ NC-3300, Aerosil™ R202, Eckagranules™ PK51/31, Epon™ 1001CX, Propylene Carbonate/EEP
(D) Epoxy-Polyamide	Epon™ 828, Epicure® 3115, Aerosil™ R202, Eckagranules™ PK51/31, Anti Terra™U, Aromatic solvent Xylol

* Adapted from: Bayer Desmodure™ E-23A Moisture Cure Urethane, Zinc Rich Primer, Formulation # 294-35, which is hereby incorporated by reference.

Example 3 -- Mg-rich Coatings Formulation and Characterization Studies

A. Critical Pigment Volume Concentration Estimates for Mg-rich Primers

Critical pigment volume concentration (CPVC) is a function of the random dense packing efficiency of the pigment plus adsorbed layer thickness (d), which must be experimentally determined. This has been discussed extensively in the literature and a recent review (Bierwagen et al., "Recent Studies of Particle Packing in Organic Coatings," Prog. Organic Coatings, 35:1-10 (1999), which is hereby incorporated by reference) considers new developments. The procedure for obtaining CPVCs for these Mg-rich systems is described as follows. Two magnesium powders, Eckagranules™ PK31 with a mean

particle size distribution (PSD) of 30 μ m, and
Eckagranules™ PK51 with a mean PSD of 70 μ m (see Figure
3A), were used as received, and mixed at a 52%-PK31:
48%-PK51 volume. A 52:48 volume mix of the two powders
5 was found to yield a higher bulk density value than that
of either powder alone. The critical pigment volume
concentration (CPVC) of the primers was first
approximated by obtaining a resin /powder rub-up value
with Aerosil™ R202 at 2% vol. on total pigment, and the
10 final CPVC was calculated from PSDs, provided by Eckart
GmbH, for the three pigments, assuming spherical
geometry, combined with the experimentally determined
resin rub-up values. Figure 3B shows calculated CPVCs
from the ternary diagram for the three-pigment mixture.
15 The volume fraction coordinates (PK31 = 0.51, PK51= 0.47,
and R202 = 0.02) yields a theoretic CPVC value in the
region where PVC = 0.475, which corroborates the
experimentally surmised CPVC, ascertained from the EIS
data in Figures 2A and 2B.

20

B. Characterization of Mg-Rich Coatings Properties

1. Panel and Film Preparation. The primers
from materials in Table 2 were applied to 6" x 3" Al 2024
T3 Q-panels™, scrubbed with a Scotch Brite™ pad, rinsed
25 and degreased with ethyl-3-ethoxypropionate ("EEP"), then
immersed in a 10% phosphoric acid solution for 60 seconds
and rinsed with distilled water. Al panels were surface
modified according to methodology described in Walker,
"Organo Silanes as Adhesion Promoters for Organic
30 Coatings," Journal of Coatings Technology, 52(670): 49-61
(1980), which is hereby incorporated by reference.
Mg-rich coatings were applied with a touch-up spray gun,
and the coatings were cured at 35°C for 14 days. Primed

panels were subsequently top coated with Extended Lifetime™ Topcoat. The average film thickness (FT) ascertained from SEM and EDAX images reveal primer film thickness to be estimated at about 50 ± 20 microns with
5 topcoat film thickness estimated at about 100 ± 40 microns.

2. Micrographs SEM and EDAX. Coated samples were assembled on aluminum mounts and coated with gold using a Technics Hummer II sputter coater. SEM and EDAX
10 images were obtained using a JEOL JSM-6300 Scanning Electron Microscope. X-ray information was obtained by a ThermoNoran EDX detector using a VANTAGE Digital Acquisition Engine. EDAX cross sections of the four 50
PVC Mg-rich primers, with pigmentary Mg X-ray
15 fluorescence (XRF) counts in red, and Silicon XRF counts, in blue, demonstrated the general alignment of Mg powder at the Al interface and pigment distribution in the polymer matrix which is thought to be related to its dispersion in the coating's polymer matrix and ultimately
20 to its degree of effectiveness as a coating for corrosion control.

C. Testing of Mg-rich Coatings

1. Mechanical Properties of Mg-Rich Coatings.
25 Tensile properties were measured according to (ASTM D 2370-82) using an Instron™ model 5542 with Merlin (2) software. DMTA measurements were made with a Rheometrics model 3-E dynamic mechanical analyzer.

2. Flammability Testing of Mg-Rich Coatings.
30 Six-inch strips were cut from top-coated Mg-rich Al panels and subjected to a modified flammability test, referenced in document IPC-SM840B (International Printed Circuit), and also described in U.L.-94 flammability

specification using a Bunsen burner, with a tube length of 4 inches, I.D. of 0.37 inches with methane gas at equivalent 1000 BTU/ft³. A propane torch with flame temperature 1120°C (2048°F) was applied for thirty seconds to the backside of the aluminum panel covered with coatings. This test was further modified by scribing an X over the face of each panel to directly expose magnesium metal in the coating to air/oxygen.

3. Exposure Testing. ^{Prohesion™} exposure was performed according to ASTM D5894-96. Top-coated Mg-rich panels were prepared by covering panel backside and edges with 3M electroplater's tape, and edges were then sealed with a 2-K industrial epoxy from Aldrich. Topcoated panels were scribed through the surface of the coating with a carbide tip glass scribe where an X pattern was formed, thus exposing the Al surface.

4. Electrochemical Impedance Spectroscopy (EIS). The corrosion protection properties of primed panels were evaluated by EIS. The experimental set-up consisted of a three-electrode cell containing 40 ml of 3.0 wt% NaCl aqueous solution, open to air, held at room temperature -22°C / 72°F. A saturated calomel electrode (SCE) was used as the reference electrode and a stainless steel plate served as the counter-electrode. All measurements were performed at the open circuit potential of the system. EIS measurements were performed with a Gamry PC-4/300™ electrochemical measurement system with potentiostat-galvanostat. Impedance spectra were recorded with a frequency sweep from 0.01Hz to 10kHz; the amplitude of the signal perturbation was 10mV (rms); and Gamry 3.1 Framework™ software was used to analyze the data in Bode plot form. These results are presented in Example 4.

Example 4 -- Test Results

A. Viscoelastic Properties of Coating Polymers

Table 3 gives the measured viscoelastic properties for five polymer systems: T_g , elastic storage modulus E' (minimum), and calculated crosslink density. The significant differences in reported glass transition temperatures are assumed to be related to the individual coatings chemical properties at full cure. Crosslink density was calculated from (E'): ($T=T_g+50^\circ\text{C}$) at which the material is in the rubbery state; where ν_e is the elastically effective crosslink density: $\nu_e = 3E'/RT(T+T_g)$.

TABLE 3
Viscoelastic Properties of Polymer Films

Polymer film	T_g ($^\circ\text{C}$)	Crosslink density (mol/cm^3)	E' (Pa) minimum
N3300 (MC-PUR)	122	3.4×10^{-3}	2.6×10^7
E23A MC-PUR	159	5.8×10^{-4}	6.6×10^5
Epoxy-polyamide	65	2.1×10^{-3}	2.0×10^7
Hyb-N3300	96	1.3×10^{-3}	1.3×10^7
Hyb-E23-A	100	6.9×10^{-4}	6.9×10^6

The results in Table 3 also suggest the difference in the chemical composition of crosslinks formed may lead to observed differences in T_g s. According to Hale and Macosko (Hale et al., Macromolecules, 24:2610 (1991), which is hereby incorporated by reference), changes in T_g arise both from disappearance of chain ends and the formation of chemical crosslinks that yield

elastically effective chain density at higher levels of branching.

B. Mechanical Properties of Mg-Rich Coatings

5 Table 4 gives the measured tensile properties of coating polymer films. Tensile tests were conducted on coating polymer film strips with no visible voids. Mechanical properties in Table 4 show an improvement in both of the hybrid system's tensile strength and tensile
10 modulus properties over their parent materials. The tensile modulus is known to be a better indication of a film's mechanical properties, as its measurement is less defect dependent than the film's tensile strength. A high tensile modulus also suggests that the material is
15 more elastic which implies higher degree of cure or conversion.

TABLE 4
Mechanical Properties of Polymer Films

20

Polymer film	Elongation at break (%)	Tensile strength (MPa)	Tensile modulus (MPa)
N3300 (MC-PUR)	8.0 \pm 0.1	45 \pm 7	1250 \pm 90
E23A MC-PUR	5.0 \pm 0.3	25 \pm 6	825 \pm 110
Epoxy-polyamide	18.0 \pm 0.1	5 \pm 0.9	150 \pm 50
25 Hyb-N3300	6.0 \pm 0.3	56 \pm 9	1800 \pm 50
Hyb-E23-A	5.5 \pm 0.2	50 \pm 5	1500 \pm 50

Both of the hybrid silane modified epoxy-urea/urethane
30 analogs show lower T_g s than the parent materials with no significant difference in crosslink density, suggesting

the presence of (-N-R-Si-O-Si-R-N-) bonds throughout the IPN matrix.

C. Flammability

5 Flame retardant (FR) coatings describe coatings that delay ignition and hinder flame spread. The common test method for evaluating flammability is the Limiting Oxygen Index (LOI) test (ASTM D 2863), a material is normally considered as flammable if the LOI is less than
10 26. These coatings were all coated with a fluorinated ELTTM topcoat that may have contributed in some measure to the coatings non-flammability (Figures 4A-4D). The most often reported parameter associated with coating flammability is the material's limiting oxygen index
15 (LOI) value. Epoxy/polyamine systems vary from a low of 24 to a high of 32 for silane modified ceramer epoxies while fluorinated polyurethanes are rated up to 50. Another contributing factor to improved non-flammability is the presence of the isocyanurate linkage. HMDI has
20 been reported to possess an inherently higher thermal stability than that of other urethane linkages, such as MDI, as the latter is reported to dissociate at about 200°C. In general, flammability decreases as the proportion of isocyanurate trizine ring increases.

25

D. Accelerated Exposure

 ProhesionTM exposure in dilute Harrison solution, (NH₄SO₄) acid rain conditions, resulted in Mg-rich coatings with conventional binders maintaining
30 clean scribes up to ~ 1,000 hours, in contrast with those coatings formulated with hybrid binders that realized clean scribes up to 3,000 hours and showed signs of failure at 5,000 hours. The integrity of the primer

vehicle appears to be the main issue associated with improved corrosion control in these systems. All samples of Mg-rich primer coatings gave better performance in this exposure than the standard chromate based system with similar topcoat, and, as seen in Figures 5A-5E, the best performing of these Mg-rich systems after 4,800 hours visibly performed better than a non-pigmented primer/ELT™ system (Figure 5E) after 1,800 hours exposure.

10

Example 5 -- Extended Electrochemical Study

A. Electrochemical Studies of Mg-Rich Coatings

EIS studies of the Mg-rich primer at 50%PVC (above CPVC) with topcoat, under conditions of high, neutral, and low pH were performed. An EIS test method was used that involved subjecting circularly scribed Mg-rich coatings to immersion varying the pH conditions in 3% NaCl solutions. This method was utilized to help differentiate among the various primer formulations developed in this work. Subjecting the system to acid (pH=2.8), neutral, and basic (pH= 2.0) 3% NaCl immersion under scribed conditions allowed comparison of the formulations in a wide range of exposure conditions. The Mg-rich Al 2024 T-3 panels were topcoated with Deft 99 GY-001 ELT™, a very chemically resistant coating, scribed, and then subjected to continuous immersion. Three coating systems were evaluated as Mg-rich coatings, two commercially available off-the-shelf products and one hybrid silane modified epoxy-urea described hereinabove. Briefly, the three coating systems were:

1. Moisture cure (MC-PUR) aromatic polyisocyanate, Desmodur™ E23A, polyurea.

2. Epoxy/polyamine consisting of Epon™ 828 with a Mannich base polyalkylamine curative Epicure™ 3251.

3. Hybrid silane modified epoxy-urea consisting of Epon 1001 and Desmodur™ N3300 aliphatic polyisocyanate,
5 Silquest™ A-1120 silane (N-β-(aminoethyl)-γ-aminopropyl trimethoxysilane) (see Table 2, formulation C).

B. Experimental Setup

Cylindrical electrode cells were mounted over
10 samples with 1.0 cm diameter circular scribes cut through the coating exposing Al 2024 T-3 surface. Cylinders were filled with electrolytes of the following compositions:

(1) Basic at 3% weight NaCl adjusted to pH= 12.0 with NaOH; (2) Acidic at 3% weight NaCl adjusted to pH=2.8
15 with HCl; (3) Neutral at 3.0% weight NaCl at pH=6.2.

Impedance measurements were carried out over an 11-day time period pH adjusted at each test interval using the EIS testing methodology described in Example 3.

Unscribed topcoated films were also examined
20 electrochemically to try to determine what is happening in undamaged coatings due to immersion in neutral 3% NaCl solution.

C. Results

25 The visual results from the scribed exposure tests (Figures 6A-6C (basic conditions pH=12) and 7A-7C (acidic conditions pH=2.8)) indicated that, under conditions of high and low pH, the Mg-rich coating formulated with traditional coating polymers, i.e.,
30 MC-PUR and the epoxy/Schiff base have weakness at these pH extremes. At high pH=12, samples A (E23, MC-PUR) and C (hybrid epoxy-Mannich base) blistered after immersion exposure. At low pH=2.8, immersion exposure caused the

film to disbond and lift from the substrate. The amino-silane modified hybrid polymer matrix (sample B) provides a much more pH resistant system in an Mg-rich coating in adhesion and reactivity than the more
5 traditional polymers. No visible difference in scribed samples after 11-days testing suggests conditions at neutral pH=6.2 yield higher stability at the primer coating/interface.

10 Example 6 -- Discussion of Results

A. Dynamic and Mechanical Property Results

Viscoelastic DMTA measurements of polymer films revealed that hybrid silane modified epoxy-urea/urethanes
15 displayed lower T_g s at equal crosslink densities to those of the parent materials suggesting the formation of bulk (-N-R-Si-O-Si-R-N-) bonds throughout the IPN matrix. In addition, the tensile properties of the silane modified epoxy-urea hybrids were better than their parent
20 materials.

B. Flammability Results

There was no observed difference in flammability with respect to PVC for any of the four
25 systems tested. The two conventional coatings, polyurea (MC-PUR) and (epoxy/polyamide) disbanded, liquefied, and incinerated with subsequent rapid magnesium incineration. The Hybrid-E23A, MDI, did not liquefy nor disbond, but formed a limited amount of char, without incineration of
30 the magnesium metal. The Hybrid N3300, aliphatic, containing N-alkylisocyanurate did not char nor did the Mg incinerate. It may be surmised that this Mg-rich coating was more covalently bonded to the Al substrate

through the (-Al-O-Si-) linkages and that this additional covalent bonding may have exerted a positive influence on decreasing the coating's flammability.

5 C. Prohesion™ Exposure Results

Results of ASTM D5894-96 performed on the four coating systems with varying pigmentary Mg content from 43, 46, and 50% PVC, showed no clear trend in PVC as a function of exposure in dilute Harrison's solution.

10 However, the 50% PVC samples in the hybrid formulations did perform slightly better. The silane modified epoxy-MDI hybrid system performed best reaching about 5,000 hours before failure, while the aliphatic silane modified epoxy-HMDI hybrid coatings failed over a range
15 of 3,000-3,400 hours. The two other conventional Mg-rich coating systems: polyurea (MC-PUR) and epoxy/polyamide, failed over a time period between 2,000 to 2,600 hours with no clear difference between the two conventional off-the-shelf coating systems.

20

Although the invention has been described in detail for the purpose of illustration, it is understood that such detail is solely for that purpose, and variations can be made therein by those skilled in the
25 art without departing from the spirit and scope of the invention which is defined by the following claims.

WHAT IS CLAIMED:

1 1. A method of treating a metal to improve the
2 metal's corrosion resistance, said method comprising:
3 applying, to the surface of the metal, a
4 coating which comprises magnesium powder and which is
5 substantially free of chromium.

1 2. A method according to claim 1, wherein the
2 metal is aluminum or an aluminum alloy.

1 3. A method according to claim 2, wherein the
2 metal is an aluminum alloy.

1 4. A method according to claim 3, wherein the
2 aluminum alloy is a copper-containing aluminum alloy.

1 5. A method according to claim 4, wherein the
2 copper-containing aluminum alloy is Al 2024 T-3.

1 6. A method according to claim 4, wherein the
2 copper-containing aluminum alloy is Al 7075 T-6.

1 7. A method according to claim 1, wherein the
2 magnesium powder comprises a mixture of a first magnesium
3 particle powder and a second magnesium particle powder,
4 wherein the first magnesium particle powder and a second
5 magnesium particle powder have substantially different
6 mean particle size distributions, wherein the mixture's
7 bulk density is greater than the first magnesium particle
8 powder's bulk density, and wherein the mixture's bulk
9 density is greater than the second magnesium particle
10 powder's bulk density.

1 8. A method according to claim 7, wherein the
2 first magnesium particle powder has a mean particle size
3 distribution of from about 25 μ m to about 35 μ m and wherein
4 the second magnesium particle powder has a mean particle
5 size distribution of from about 65 μ m to about 75 μ m.

1 9. A method according to claim 8, wherein the
2 first magnesium particle powder has a mean particle size
3 distribution of about 30 μ m and wherein the second
4 magnesium particle powder has a mean particle size
5 distribution of from about 70 μ m.

1 10. A method according to claim 8, wherein the
2 mixture contains first magnesium particle powder and
3 second magnesium particle powder in a volume ratio of
4 from about 40:60 to about 60:40.

1 11. A method according to claim 8, wherein the
2 mixture contains first magnesium particle powder and
3 second magnesium particle powder in a volume ratio of
4 from about 45:55 to about 55:45.

1 12. A method according to claim 8, wherein the
2 mixture contains first magnesium particle powder and
3 second magnesium particle powder in a volume ratio of
4 from about 50:50 to about 55:45.

1 13. A method according to claim 12, wherein
2 the first magnesium particle powder has a mean particle
3 size distribution of about 30 μ m and wherein the second
4 magnesium particle powder has a mean particle size
5 distribution of from about 70 μ m.

1 14. A method according to claim 13, wherein
2 the mixture contains first magnesium particle powder and
3 second magnesium particle powder in a volume ratio of
4 about 58:42.

1 15. A method according to claim 1, wherein the
2 coating further comprises a polymeric binder.

1 16. A method according to claim 15, wherein
2 the polymeric binder comprises a polyisocyanate
3 prepolymer and an epoxy prepolymer.

1 17. A method according to claim 16, wherein
2 the polyisocyanate prepolymer is an aliphatic
3 polyisocyanate prepolymer.

1 18. A method according to claim 16, wherein
2 the polyisocyanate prepolymer is an aromatic
3 polyisocyanate prepolymer.

1 19. A method according to claim 16, wherein
2 the polymeric binder comprises a polyisocyanate
3 prepolymer and an epoxy prepolymer and wherein said
4 method further comprises contacting the polymeric binder
5 with a crosslinker.

1 20. A method according to claim 19, wherein
2 the crosslinker is a silanated tetrahydroquinoxalinol.

1 21. A method according to claim 19, wherein
2 the crosslinker is a 7-phenyl-1-[4-(trialkylsilyl)-
3 alkyl]-1,2,3,4-tetrahydroquinoxalin-6-ol.

1 22. A method according to claim 19, wherein
2 the crosslinker is a 7-phenyl-1-[4-(trialkylsilyl)-
3 butyl]-1,2,3,4-tetrahydroquinoxalin-6-ol.

1 23. A method according to claim 1, wherein,
2 prior to said applying, said method further comprises:
3 contacting the metal surface with an amine-
4 containing organo-silane.

1 24. A method according to claim 23, wherein
2 the amine-containing organo-silane is (N- β -(aminoethyl)-
3 γ -aminopropyltrimethoxysilane.

1 25. A method according to claim 15, wherein
2 the polymeric binder is a silane modified epoxy
3 isocyanate hybrid.

1 26. A method according to claim 15, wherein
2 the polymeric binder is a polymeric material containing
3 polyurea, polyurethane, epoxy-amine, and organo-silane
4 linkages.

1 27. A method according to claim 1, wherein the
2 coating further comprises a rare earth metal or anion
3 thereof.

1 28. A method according to claim 27, wherein
2 the rare earth metal is cerium.

1 29. A coating composition comprising:
2 magnesium powder; and
3 a silane modified epoxy isocyanate hybrid
4 polymer or prepolymer.

1 30. A coating composition according to claim
2 29, wherein said magnesium powder is substantially
3 uniformly dispersed in said silane modified epoxy
4 isocyanate hybrid polymer or prepolymer.

1 31. A coating composition according to claim
2 29, wherein said magnesium powder comprises a mixture of
3 a first magnesium particle powder and a second magnesium
4 particle powder, wherein the first magnesium particle
5 powder and a second magnesium particle powder have
6 substantially different mean particle size distributions,
7 wherein the mixture's bulk density is greater than that
8 of the first magnesium particle powder's bulk density,
9 and wherein the mixture's bulk density is greater than
10 that of the second magnesium particle powder's bulk
11 density.

1 32. A coating composition according to claim
2 31, wherein the first magnesium particle powder has a
3 mean particle size distribution of from about 25 μ m to
4 about 35 μ m and wherein the second magnesium particle
5 powder has a mean particle size distribution of from
6 about 65 μ m to about 75 μ m.

1 33. A coating composition according to claim
2 32, wherein the first magnesium particle powder has a
3 mean particle size distribution of about 30 μ m and wherein
4 the second magnesium particle powder has a mean particle
5 size distribution of from about 70 μ m.

1 34. A coating composition according to claim
2 32, wherein the mixture contains first magnesium particle

3 powder and second magnesium particle powder in a volume
4 ratio of from about 40:60 to about 60:40.

1 35. A coating composition according to claim
2 32, wherein the mixture contains first magnesium particle
3 powder and second magnesium particle powder in a volume
4 ratio of from about 45:55 to about 55:45.

1 36. A coating composition according to claim
2 32, wherein the mixture contains first magnesium particle
3 powder and second magnesium particle powder in a volume
4 ratio of from about 50:50 to about 55:45.

1 37. A coating composition according to claim
2 36, wherein the first magnesium particle powder has a
3 mean particle size distribution of about 30 μ m and wherein
4 the second magnesium particle powder has a mean particle
5 size distribution of from about 70 μ m.

1 38. A coating composition according to claim
2 37, wherein the mixture contains first magnesium particle
3 powder and second magnesium particle powder in a volume
4 ratio of about 58:42.

1 39. A coating composition according to claim
2 29, wherein said silane modified epoxy isocyanate hybrid
3 polymer or prepolymer comprises a polyisocyanate
4 prepolymer, an epoxy prepolymer, and a silanated
5 tetrahydroquinoxalinol crosslinker or a polymerization
6 product thereof.

1 40. A coating composition according to claim
2 39, wherein the polyisocyanate prepolymer is an aliphatic
3 polyisocyanate prepolymer.

1 41. A coating composition according to claim
2 39, wherein the polyisocyanate prepolymer is an aromatic
3 polyisocyanate prepolymer.

1 42. A coating composition according to claim
2 39, wherein the silanated tetrahydroquinoxalinol
3 crosslinker is a 7-phenyl-1-[4-(trialkylsilyl)-alkyl]-
4 1,2,3,4-tetrahydroquinoxalin-6-ol.

1 43. A coating composition according to claim
2 39, wherein the silanated tetrahydroquinoxalinol
3 crosslinker is a 7-phenyl-1-[4-(trialkylsilyl)-butyl]-
4 1,2,3,4-tetrahydroquinoxalin-6-ol.

MAGNESIUM RICH COATINGS

ABSTRACT OF THE DISCLOSURE

Disclosed are methods of treating a metal to improve the metal's corrosion resistance. The method includes applying, to the surface of the metal, a coating which comprises magnesium powder and which is

5 substantially free of chromium. The present invention also relates to a coating composition that includes magnesium powder and a silane modified epoxy isocyanate hybrid polymer or prepolymer. The inventors have found that corrosion resistance (as determined by PROHESION™

10 exposure in accordance with ASTM D5894-96, which is hereby incorporated by reference) in excess of 3,000 hours on 2024 T-3 aluminum alloy can be achieved with the methods and coating compositions of the present invention.

Figure 1A

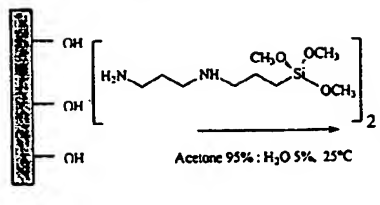


Figure 1B

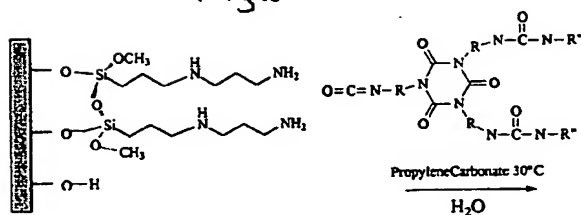


Figure 1C

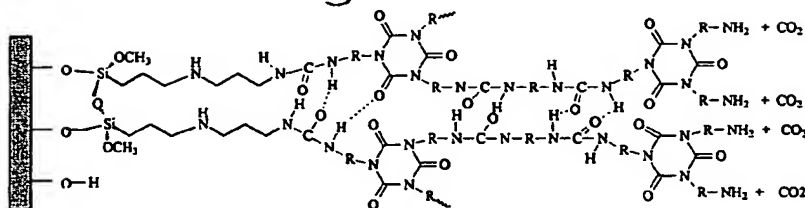
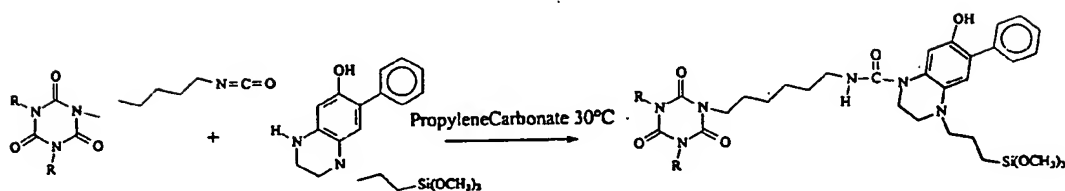


Figure 1D



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Figure 2A

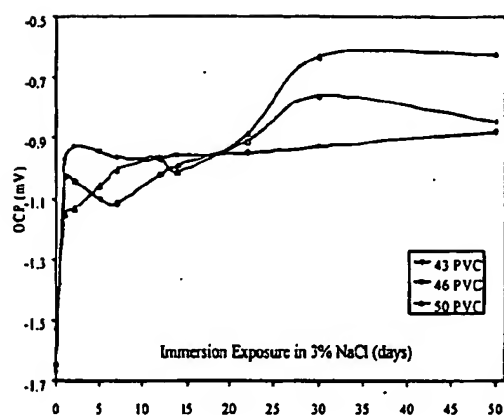


Figure 2B

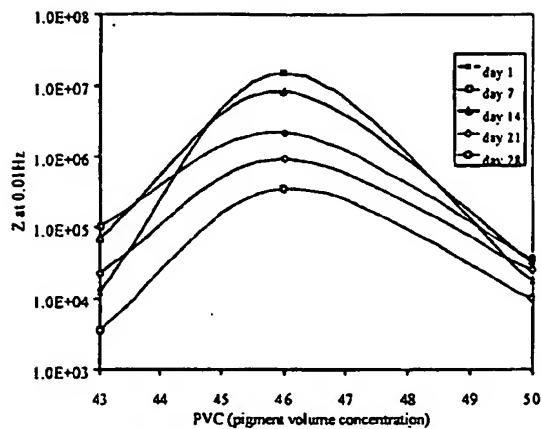


Figure 3A

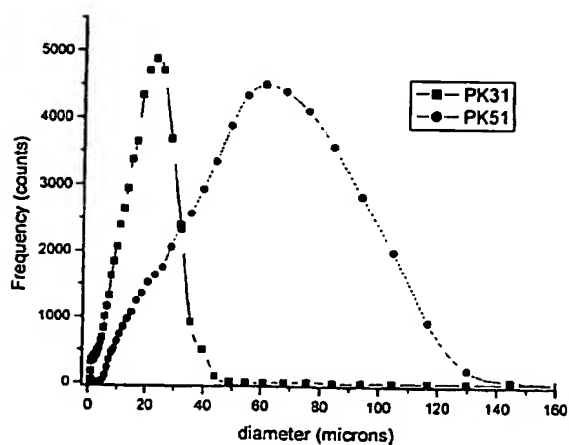
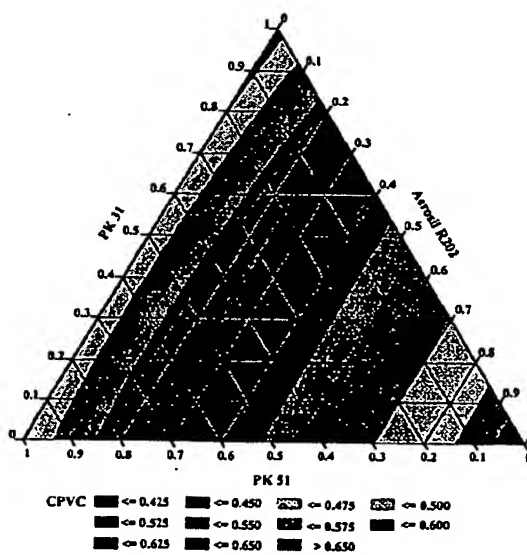


Figure 3B



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Figure 4

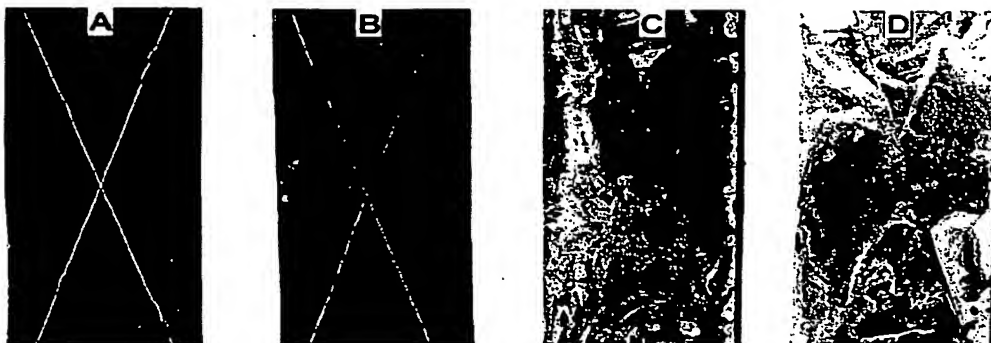
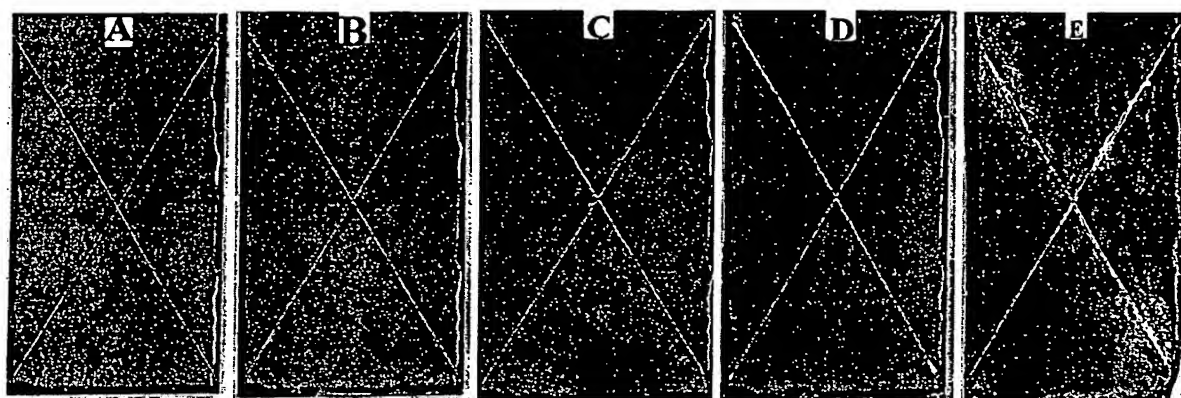


Figure 5



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Figure 6

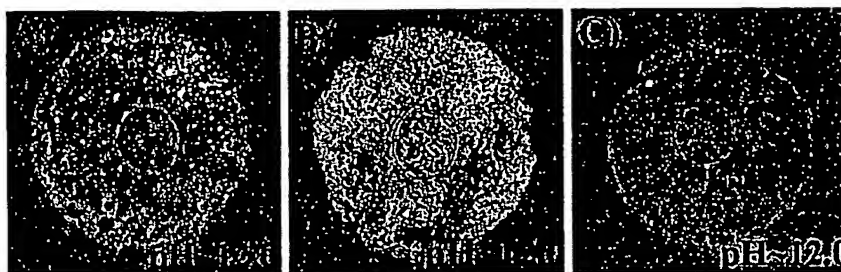
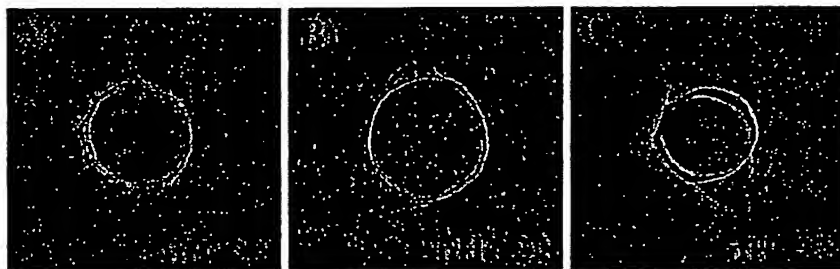


Figure 7



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